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The Planck mission

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Abstract

We present an overview of the European Space Agency's Planck mission, its scientific objectives and the main elements of its technical design. The current programmatic status of Planck within ESA's Scientific Programme, implementation plans, and near-term milestones are also outlined.

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1. Science

Planck (<http://astro.estec.esa.nl/Planck>) is a space observatory designed to image the temperature anisotropies of the Cosmic Microwave Background (CMB) over the whole sky, with unprecedented sensitivity ($\Delta T/T \sim 2 \times 10^{-6}$) and angular resolution (better than 10 arcmin). Planck will provide a major source of information relevant to several cosmological and astrophysical issues, such as testing theories of the early universe and the origin of cosmic structure.

The ability to measure to high accuracy the angular power spectrum of the CMB fluctuations will allow the determination of fundamental cosmological parameters such as the density parameter (Ω_0), and the Hubble constant H_0 , with an uncertainty of order a few percent (e.g., Bersanelli et al., 1996, Bond et al., 1997; Efstathiou and Bond, 1999), Planck will not only measure the temperature fluctuations of the CMB, but also its polarization state. This measurement (e.g., Seljak, 1997) will not only yield new scientific results, but will also help to analyze the CMB temperature anisotropies (in particular by resolving degeneracies in the estimation of some cosmological parameters; see Zaldarriaga et al., 1997; Kamionkowski and Kosowsky, 1998).

In addition to the main cosmological goals of the mission, the Planck sky survey will be used to study in detail the very sources of emission which “contaminate” the signal due to the CMB, and will result in a wealth of information on the properties of extragalactic sources, and on the dust and gas in our own galaxy (e.g., de Zotti et al., 1999). One specific notable result will be the measurement of the Sunyaev-Zeldovich (SZ) effect in many thousands of galaxy clusters (Aghanim et al., 1997; Pointecouteau et al., 1998).

2. Payload

The scientific prescription which will allow Planck to meet its ambitious objectives calls for:

- an offset telescope with a physical aperture of size ~ 1.5 m to achieve the angular resolution.
- state-of-the-art broadband detectors covering the range ~ 25 –1000 GHz, to achieve the required sensitivity and the ability to remove foreground sources of emission.
- a survey with all-sky coverage carried out from a far-Earth orbit.
- extreme attention to rejection of unwanted systematic effects.

To achieve this prescription a payload was conceived for Planck (Bersanelli et al., 1996) consisting of three basic components: (1) a telescope and baffling system,

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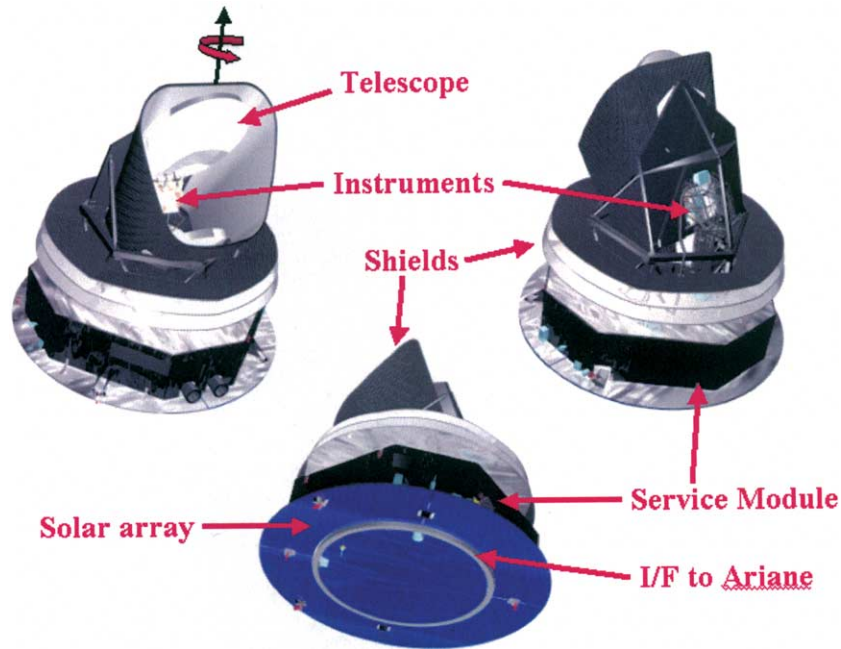


Fig. 1. A conceptual view of the arrangement of the main elements of the Planck payload. The instrument focal plane unit contains both low and high frequency detectors. The function of the large shield surrounding the telescope is to control the far sidelobe level of the radiation pattern as seen from the detectors. Specular cones thermally decouple the Service Module (located below the Payload Module), and allow the payload environment to reach its nominal temperature of ~ 50 K. The satellite's spin axis is indicated; the Sun is always kept in the anti-spin direction (perpendicular to the solar array). The satellite views are courtesy of Alcatel Space (Cannes).

63 providing the angular resolution and rejection of stray-
 64 light; (2) a Low Frequency Instrument (or LFI) – an
 65 array of tuned radio receivers, based on HEMT ampli-
 66 fiers, covering the frequency range 25–80 GHz¹, and
 67 operated at a temperature of 20 K; and (3) a High
 68 Frequency Instrument (HFI), consisting of an array of
 69 bolometers operated at 0.1 K and covering the fre-
 70 quency range 90–1000 GHz.

71 The major elements of the Planck payload and their
 72 disposition on the spacecraft can be seen in Fig. 1.

73 3. Programmatic

74 Planck was selected as the third Medium-Sized Mis-
 75 sion (M3) of ESA's Horizon 2000 Scientific Programme
 76 (<http://sci.esa.int/home/ourmissions/index.cftn>). Planck
 77 was formerly called COBRAS/SAMBA. After the mis-
 78 sion was selected and approved (in late 1996), it was
 79 renamed in honor of the German scientist Max Planck
 80 (1858–1947), Nobel Prize for Physics in 1918. Planck
 81 will be launched together with ESA's Far-Infrared and
 82 Submillimeter Space Observatory (Herschel) in Febru-
 83 ary of 2007.

¹ Funding shortfalls within the LFI Consortium have forced to delete the originally present 100 GHz channel from the LFI baseline configuration.

Starting in 1993, a number of technical studies laid
 the basis for the issue in September 2000 of an Invitation
 to Tender (ITT) to European industry for the procure-
 ment of the Herschel and Planck spacecraft. From the
 submitted proposals, a single prime contractor, Alcatel
 Space (France), was selected in early 2001 to develop
 both Herschel and Planck spacecrafts. Alcatel Space is
 supported by two major subcontractors: Alenia Spazio
 (Torino) for the Service Module, and Astrium GmbH
 (Friedrichshafen) for the Herschel Payload Module; and
 by many other industrial subcontractors from all ESA
 member states. The detailed definition work began in
 June 2001, and is very advanced at the time of writing. A
 view of the current design of Planck is shown in Fig. 1.

Planck is a survey-type project which is being devel-
 oped and operated as a Principal Investigator mission.
 In early 1999, ESA selected two Consortia of scientific
 institutes to provide the two Planck instruments: the
 Low Frequency Instrument will be developed and deliv-
 ered to ESA by a Consortium led by Reno Mandolesi
 of the Istituto di Astrofisica Spaziale (CNR) in Bolo-
 gna (Italy); similarly, the High Frequency Instrument
 will be provided to ESA by a Consortium led by Jean-
 Loup Puget of the Institut d' Astrophysique Spatiale
 (CNRS) in Orsay (France). More than 40 European
 institutes, and some from the USA, are collaborating on
 the development, testing, and operation of these in-
 struments, as well as the ensuing data analysis and ex-

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112 ploitation. The capabilities of the instruments are de-
 113 scribed in Table 1.

114 In early 2000, ESA and the Danish Space Research
 115 Institute (DSRI, Copenhagen) signed an Agreement for
 116 the provision of the two reflectors that constitute the
 117 Planck telescope. DSRI leads a Consortium of Danish
 118 institutes, and has subcontracted the development of the
 119 Planck reflectors to Astrium GmbH (Friedrichshafen),
 120 who will manufacture the reflectors using state-of-the-
 121 art carbon fiber technology. At the time of writing the
 122 design of the reflectors is finished, the molds needed for
 123 the layout of the facesheets are very advanced, and the
 124 reflector development is proceeding according to plans.

125 In parallel, the instrument development has pro-
 126 ceeded largely according to schedule, in spite of a
 127 number of financial difficulties during 2002, which at the
 128 time of writing have been largely resolved. Some hard-
 129 ware subsystems are already being manufactured and
 130 tested. The first deliveries of instrument qualification
 131 models are expected in late 2003, with the flight models
 132 due in early 2005.

133 4. Scientific performance

134 The principal objective of Planck is to produce maps
 135 of the whole sky in 10 frequency channels. The currently
 136 foreseen characteristics of the two Planck instruments
 137 (see Fig. 2) are summarized in Table 1; these charac-
 138 teristics largely drive the quality of the final maps.

139 These maps will not only include the CMB itself, but
 140 also all other astrophysical foregrounds, whether ga-
 141 lactic (free-free, synchrotron or dust) or extragalactic in
 142 origin. All 10 Planck sky maps will be used to produce a
 143 single map of the Cosmic Microwave Background an-
 144 isotropies. The key that allows to reach this objective is

145 the wide spectral coverage achieved by Planck. Each
 146 astrophysical foreground has a distinct (albeit at present
 147 poorly known) spectral characteristic. Specialized data
 148 processing algorithms (e.g., Tegmark, 1997; Hobson et
 149 al., 1998; Bouchet and Gispert, 1999) will use this in-
 150 formation to iteratively extract the signal due to each
 151 foreground component, until only the CMB signal re-
 152 mains. Instrumental systematic effects (e.g., Delabrou-
 153 ille, 1998; Burigana et al., 1998; Maino et al., 1999), as
 154 well as uncertainties in the recovery of parameters
 155 characterizing the foregrounds will degrade the final
 156 noise level in the CMB maps.

157 Therefore the final scientific performance of the
 158 mission depends not only on the instrumental behavior,
 159 but also on the detailed nature of the various astro-
 160 physical foregrounds, the behavior of many systematic
 161 effects which produce spurious signals (such as stray-
 162 light), and the ability to remove these signals from the
 163 measured data by means of data processing algorithms.
 164 Current estimates of the performance of Planck are
 165 based on simulations of the measurement process (e.g.,
 166 Bersanelli et al., 1996; Bouchet and Gispert, 1999;
 167 Knox, 1999; Tegmark et al., 2000), which include such
 168 effects to the best of available knowledge, as well as of
 169 the signal extraction process. These simulations suggest
 170 that, the ability to extract the CMB signal from the
 171 measurements will be limited mainly by the background
 172 of noise originating in unresolved structure in the vari-
 173 ous foregrounds.

174 The Planck instruments are also designed to provide
 175 information on the polarization state of the CMB. Al-
 176 though simulations of the extraction of polarization
 177 from Planck are at a less sophisticated level than those
 178 dealing with temperature anisotropies, it is expected that
 179 Planck will be able to measure with good accuracy the
 180 angular power spectrum of the “E-component” of CMB 180

Table 1
 Characteristics of the Planck Payload

Telescope	1.5 m (proj. apert.) Aplanatic; Temp. ~50 K; $\epsilon_{\text{system}} \sim 1\%$								
	Shared focal plane; viewing direction offset 85° from spin axis								
Instrument	LFI			HFI					
Center frequency (GHz)	30	44	70	100	143	217	353	545	857
Detector technology	HEMT receiver arrays			Bolometer arrays					
Detector temperature	~20 K			0.1 K					
Cooling requirements	H ₂ sorption cooler			H ₂ sorption + 4K J-T stage + Dilution					
Number of unpolarized detectors	–	–	–	4	4	4	4	4	4
Number of linearly polarized detectors	4	6	12	^a	8	8	8	–	–
Bandwidth ($\Delta\nu/\nu$)	0.2	0.2	0.2	0.33	0.33	0.33	0.33	0.33	0.33
Angular resolution (arcmin)	33	24	14	9.2	7.1	5.0	5.0	5.0	5.0
Average $\Delta T/T$ per pixel (Stokes I) (12 mos, 1σ , 10^{-6} units)	2.0	2.7	4.7	2.0	2.2	4.8	14.7	147.0	6700
Average $\Delta T/T$ per pixel (Stokes U, Q) (12 mos, 1σ , 10^{-6} units)	2.8	3.9	6.7	^a	4.2	9.8	29.8	–	–

^aThe possibility of adding polarization capability is being investigated.

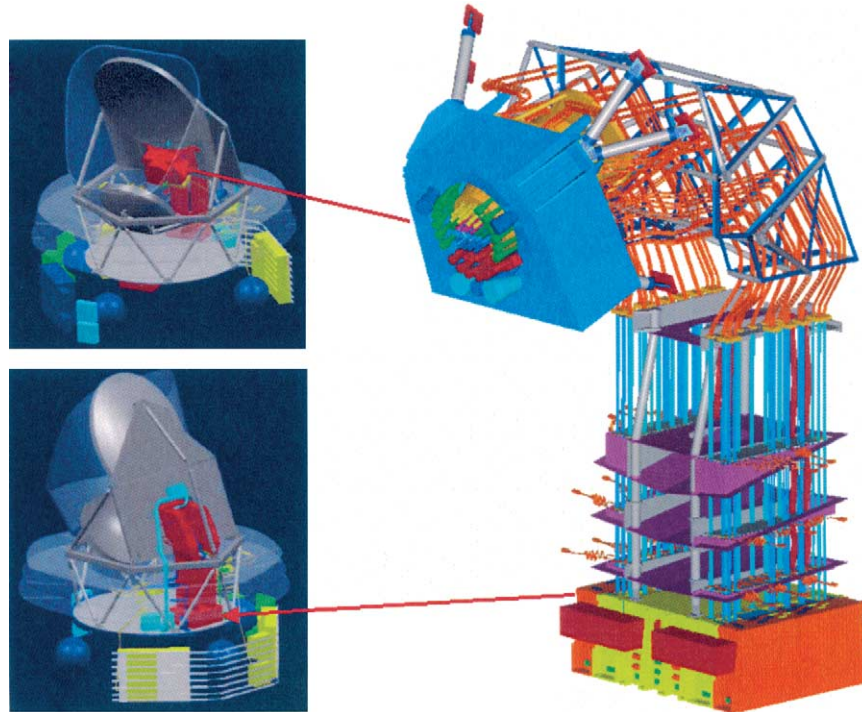


Fig. 2. On the right (top) is shown the combined focal plane unit formed by the LFI and HFI (see also Note 1 on p. 2) ; the colors on each horn correspond to frequency channels. The HFI focal plane is inserted into the ring formed by the LFI horns, and includes thermal stages at 20, 4, 2 and 0.1 K. The LFI horns and detectors are cooled to 20 K, and are attached by waveguides to a lower unit containing back-end amplifiers; a complex mechanical structure stiffens the assembly. On the left is shown the manner in which the various instrument units are distributed throughout the Payload and Service Modules. In general, all cryogenic units are located in the Payload Module and all warm units (electronics, coolers) are located in the Service Module. The figures are courtesy of Alcatel Space (Cannes) and Laben (Milano).

181 polarization (Hu and White, 1997), with the consequent
 182 improved estimation of cosmological parameters (Bou-
 183 chet et al., 1999; Prunet et al., 2000).

184 Naturally, as a by-product of the extraction of the
 185 CMB, Planck will also yield all-sky maps of all the
 186 major sources of microwave to far-infrared emission,
 187 opening a broad expanse of astrophysical topics to
 188 scrutiny. In particular, the physics of dust at long
 189 wavelengths and the relative distribution of interstellar
 190 matter (neutral and ionized) and magnetic fields will be
 191 investigated using dust, free-free and synchrotron maps.
 192 In the field of star formation, Planck will provide a
 193 systematic search of the sky for dense, cold condensa-
 194 tions which are the first stage in the star formation
 195 process. One specific and local distortion of the CMB
 196 which will be mapped by Planck is the SZ effect arising
 197 from the Compton interaction of CMB photons with the
 198 hot gas of clusters of galaxies. The very well defined
 199 spectral shape of the SZ effect allows it to be cleanly
 200 separated from the primordial anisotropy. The physics
 201 of gas condensation in cluster-size potential wells is an
 202 important element in the quest to understand the
 203 physics of structure formation and ultimately of galaxy
 204 formation.

205 Therefore, even though that is not its primary scien-
 206 tific objective, Planck will deliver high quality all-sky

maps of all extended foreground emission components 207
 between cm and submm wavelengths. These maps will 208
 constitute a scientific product which is comparable to 209
 the IRAS and COBE-DIRBE maps at shorter wave- 210
 lengths. The high-resolution all-sky nature of the maps, 211
 coupled with the broad spectral coverage and highly 212
 accurate calibration, will present a unique opportunity 213
 to explore in a global sense all the extended emission 214
 components of our own as well as external galaxies. 215

5. Mission profile 216

Planck will be launched together with Herschel in 217
 February of 2007 by an Ariane 5 rocket from the Eu- 218
 ropean spaceport in Kourou (French Guiana). Planck 219
 and Herschel will separate immediately after launch, 220
 and each will proceed independently to different orbits 221
 around the L2 point of the Earth-Sun system. At this 222
 location, the payload can be continuously pointed in the 223
 anti-Sun direction, thus minimizing potentially confus- 224
 ing signals due to thermal fluctuations and straylight 225
 entering the detectors through the far sidelobes. 226

The transit time for Planck will be between 3 and 4 227
 months; this period will be used for commissioning of 228
 the spacecraft and instruments. The spacecraft (S/C) will 229

230 be placed into a Lissajous orbit around L2 characterized
 231 by a ~ 4 month period and a maximum elongation from
 232 L2 of about 3,80,000 km, such that the Sun–S/C–Earth
 233 angle will not exceed 15° . From this orbit, Planck will
 234 carry out two complete surveys of the full sky, for which
 235 it requires between 12 and 14 months of observing time.

236 The satellite will rotate at 1 rpm around a spin axis
 237 pointed within 10° of the Sun. The payload will always
 238 remain in the shadow of the Sun. The solar array en-
 239 sures this as long as it is inclined with respect to the Sun–
 240 S/C line by less than 10° . The Planck telescope and focal
 241 plane define a sparsely sampled field of view (FOV)
 242 approximately 8° in diameter around a reference line-of-
 243 sight which is inclined by 85° with respect to the spin
 244 axis. As the satellite rotates, the FOV will thus trace a
 245 circle of diameter 170° on the sky.

246 In order to carry out its two consecutive full-sky
 247 surveys and maintain the payload in the solar shadow,
 248 the spin axis of Planck must be displaced on the average
 249 by 1° per day in the direction defined by the orbital
 250 motion of the Earth around the Sun. This is achieved by
 251 spin axis depointing manoeuvres at regular intervals. As
 252 the spin axis is displaced, the observed circle also moves
 253 and gradually covers a large fraction of the sky.

254 Planck will dump each day to Earth within a period
 255 of 3 h the data acquired during 24 h. Observations will
 256 not be interrupted during the downlink period, and the
 257 S/C will not be reoriented towards the Earth. The te-
 258lemetry antenna is designed to have adequate gain
 259 within a 15° half-cone from the spin axis, ensuring that
 260 even at the extremes of its orbit the Planck telemetry can
 261 achieve full bandwidth.

262 6. Operations and data processing

263 The Planck spacecraft will be controlled from a
 264 dedicated Mission Operations Centre developed and
 265 operated by ESOC in Darmstadt (Germany). From
 266 there, the scientific data produced by Planck will be
 267 piped daily to two Data Processing Centres (DPCs),
 268 which will be developed and operated by the two Con-
 269 sortia selected to provide the Planck instruments.

270 In particular, the two DPCs will be responsible for:
 271 (a) daily and long term analysis of instrument health and
 272 performance; (b) daily analysis of science data; (c) all
 273 levels of processing of Planck data, from raw telemetry
 274 to deliverable scientific products.

275 The two DPCs will share a basic information man-
 276 agement infrastructure, the Planck Integrated Data and
 277 Information System (IDIS, Bennett et al., 2000). IDIS is
 278 being conceived from an object-oriented point of view,
 279 and is planned to contain five different components: (a)
 280 a Document Management Component, containing all
 281 relevant documentation; (b) a Software Management
 282 Component, encompassing the software in common

between the two Consortia; (c) a Process Coordinator 283
 Component, providing a single software environment 284
 for data processing (e.g., a data pipeline manager); (d) a 285
 Data Management Component, allowing the ingestion, 286
 efficient management and extraction of the data (or 287
 subsets thereof) produced by Planck activities; (e) a 288
 Federation layer, providing inter-connection among 289
 IDIS components (e.g., relating objects controlled by 290
 each component). 291

The main scientific products of the mission will be 292
 produced by the two DPCs jointly, and will consist of 293
 all-sky maps in ten frequency bands, which will be made 294
 publicly available one year after completion of the 295
 mission (i.e., in late 2010), together with a first genera- 296
 tion set of maps of the CMB, SZ effect, galactic emission 297
 (dust, free-free, and synchrotron), and point-source 298
 catalogs. The time series of observations (after calibra- 299
 tion and position reconstruction) will also eventually be 300
 made available as an on-line archive. 301

7. Uncited references 302

Bennett et al. (1996), Mather et al. (1999), Revenu et 303
 al. (2000) 304

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 the past years by the Planck Instrument and Reflector 311
 Provider Consortia, ESA, ESOC, and industry (princi- 312
 pally the former Matra Marconi Toulouse, and Alcatel 313
 Space Cannes). 314

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